

Mechanics of Materials

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Chapter 3

Analysis of Strain Concepts and Definitions

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Contents

Analysis of Strain, Concepts and Definitions

- 3-1 Introduction (Self read)
- **3-2 Displacement, Deformation, and Strain**
- **3-3 The State of Strain** at a Point
- **3-4 The Strain Transformation Equations for Plane Strain**
- **3-5 Principal Strains and Maximum Shearing Strains**
- **3-6 Mohr's Circle** for Plane Strain
- **3-7 Strain Measurement and Rosette**

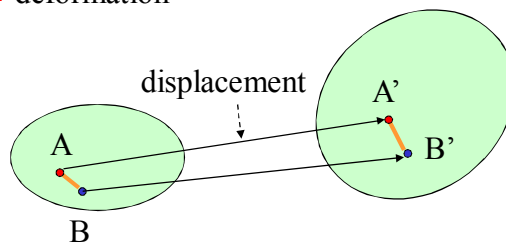
p.121

3-2 Displacement, Deformation, and Strain

■ Displacement

a vector indicating the movement of a point with respect to some coordinate system.

- rigid body motion
- deformation

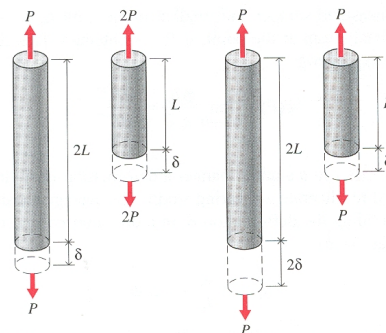


p.122

3-2 Displacement, Deformation, and Strain

■ Deformation

Change of size and/or shape of a body.



Note: The elongation induced by the same load may be different.

⇒ Need a quantitative measure of the “intensity of deformation”.

p.123

3-2 Displacement, Deformation, and Strain

■ Strain

Deformation per unit length (Intensity of a deformation)

- normal strain ϵ
- shearing strain γ

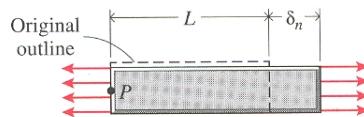
❖ Axial Strain

Average

$$\epsilon_{avg} = \frac{\delta_n}{L}$$

Local point P

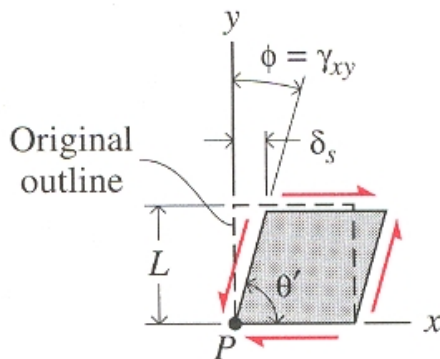
$$\epsilon(P) = \lim_{\Delta L \rightarrow 0} \frac{\Delta \delta_n}{\Delta L} = \frac{d\delta_n}{dL}$$



p.123

3-2 Displacement, Deformation, and Strain

❖ Shearing Strain



$$\gamma_{avg} = \frac{\delta_s}{L} = \tan \phi \cong \sin \phi \cong \phi$$

$$\gamma_{xy}(P) = \lim_{\Delta L \rightarrow 0} \frac{\Delta \delta_s}{\Delta L} = \frac{d\delta_s}{dL} \cong \frac{\pi}{2} - \theta'$$

Note: Usually $\delta_s/L < 0.001$, $\tan \phi \cong \sin \phi \cong \phi$.

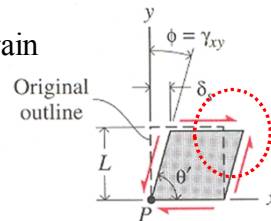
γ : decrease in the angle between two reference lines that are orthogonal in the undeformed state.

p.124

3-2 Displacement, Deformation, and Strain

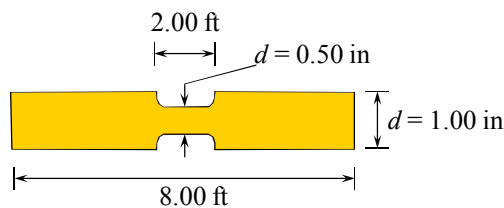
- Units of Strain : **dimensionless**
 - Normal strain: m/m (or $\mu\text{m/m}$)
 - Shear strain: rad (or $\mu\text{-rad}$). ?

- Sign Convention for Strains
 - Normal strain:
 - elongation : + tensile strain
 - contraction : - compressive strain
 - Shearing strain
 - θ' **decreases : +**



p.124

Example Problem 3-1



Given $(\epsilon_{avg})_{center} = 960 \mu \text{ in/in}$
 $\delta_{total} = 0.04032 \text{ in}$

Find: ■ $\delta_{Center} = ?$
 ■ $\epsilon_{End} = ?$

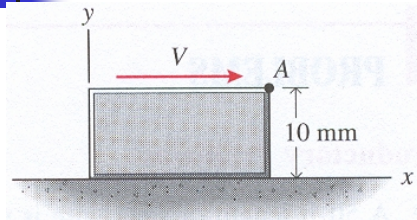
$$\delta_C = \epsilon_{avg} L = 960(10^{-6})(2)(12) = 0.02304 \text{ in} \approx 0.0230 \text{ in} = 0.585 \text{ mm}$$

$$\delta_E = \delta_{total} - \delta_C = 0.04032 \text{ in} - 0.02304 \text{ in} = 0.01728 \text{ in} = 0.439 \text{ mm}$$

$$\epsilon_E = \frac{\delta_E}{L} = \frac{0.01728}{6(12)} = 240(10^{-6}) = 240 \mu \text{ in/in} = 240 \mu \text{ mm/mm}$$

p.125

Example Problem 3-2



$$\gamma_{avg} = 1000 \mu\text{m/m}$$

Determine

■ Displacement of $A = ?$

(1) $\delta_A = 1 \mu\text{m}$

(2) $\delta_A = 10 \mu\text{m}$

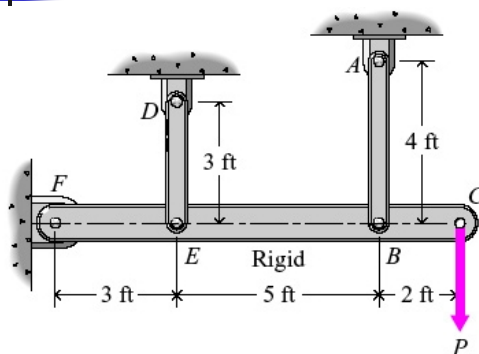
(3) $\delta_A = 100 \mu\text{m}$

(4) $\delta_A = 1000 \mu\text{m}$

$$\delta_A = \gamma_{avg} L = 1000(10^{-6})(10)\text{mm} = 0.01 \text{ mm} = 10 \mu\text{m}$$

p.125

Example Problem 3-3 (I)



$$\epsilon_{DE} = 0.0006 \text{ in/in}$$

Determine

■ $\delta_{AB} = ?$ $\epsilon_{AB} = ?$

■ $\delta_{AB} = ?$ if there is a 0.001 in clearance in the connection at B between bar AB and CF.

p.125

Example Problem 3-3

$$\vec{BB'} = -u_B \hat{i} - v_B \hat{j}$$

$$\delta_{AB} = AB' - AB = AB' - L$$

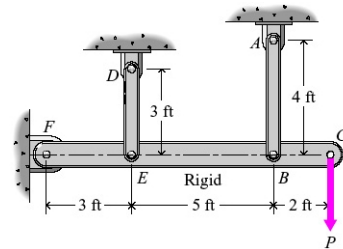
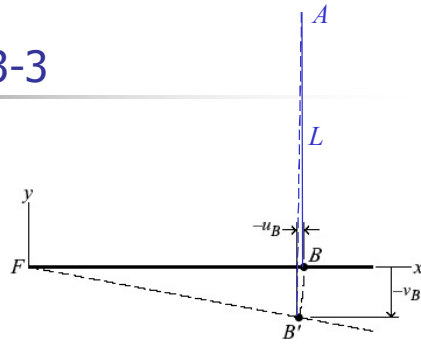
$$\delta_{AB} = \sqrt{(L + v_B)^2 + u_B^2} - L$$

$$\delta_{AB}^2 + 2\delta_{AB}L + \cancel{L^2} = \cancel{L^2} + 2v_B L + v_B^2 + u_B^2$$

Neglect squares of small displacements,

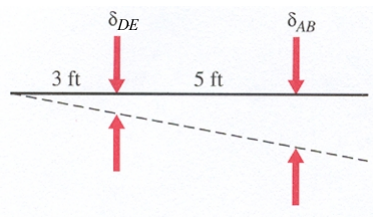
$$\cancel{\delta_{AB}^2} + 2\delta_{AB}L = 2v_B L + \cancel{v_B^2} + \cancel{u_B^2}$$

$$\Rightarrow \delta_{AB} \cong v_B \quad \text{Similarly, } \Rightarrow \delta_{DE} \cong v_E$$



p.125

Example Problem 3-3 (II)



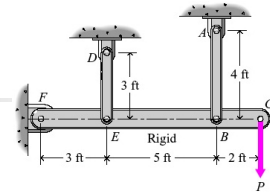
Determine

$$\delta_{AB} = ? \quad \epsilon_{AB} = ?$$

$$\delta_{DE} = \epsilon_{DE} L_{DE} = 0.0006(3\text{ft})(12\text{in}/\text{ft}) = 0.0216 \text{ in}$$

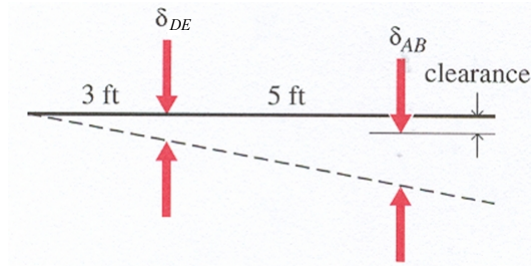
$$\delta_{AB} = \frac{8}{3} \delta_{DE} = \frac{8}{3}(0.0216\text{in}) = 0.0576 \text{ in}$$

$$\epsilon_{AB} = \frac{\delta_{AB}}{L_{AB}} = \frac{0.0576}{4(12)} = 0.001200 \text{ in}/\text{in} = 1200 \mu\text{in}/\text{in}$$



p.126

Example Problem 3-3 (III)



Determine

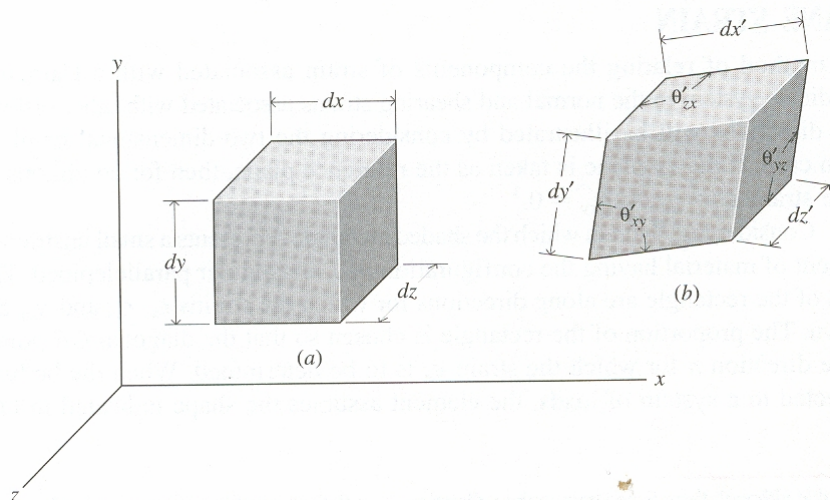
■ $\delta_{AB} = ?$ $\epsilon_{AB} = ?$

$$\delta_{AB} = \frac{8}{3}\delta_{DE} - \text{clearance} = \frac{8}{3}(0.0216) - 0.001 = 0.0566 \text{ in}$$

$$\epsilon_{AB} = \frac{\delta_{AB}}{L_{AB}} = \frac{0.0566}{4(12)} = 0.001179 \text{ in/in} = 1179 \mu\text{in/in}$$

p.129

3-3 The State of Strain at a Point

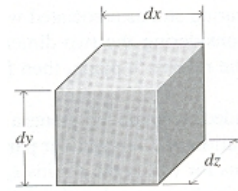


p.129

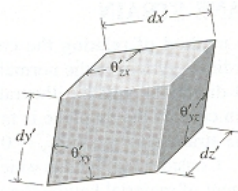
3-3 The State of Strain at a Point

- Strain components associated with the Cartesian coordinates

$$\left\{ \begin{array}{l} \varepsilon_x = \frac{dx' - dx}{dx} = \frac{d\delta_x}{dx} \\ \varepsilon_y = \frac{dy' - dy}{dy} = \frac{d\delta_y}{dy} \\ \varepsilon_z = \frac{dz' - dz}{dz} = \frac{d\delta_z}{dz} \end{array} \right. \Rightarrow \left\{ \begin{array}{l} dx' = (1 + \varepsilon_x)dx \\ dy' = (1 + \varepsilon_y)dy \\ dz' = (1 + \varepsilon_z)dz \end{array} \right.$$



$$\left\{ \begin{array}{l} \gamma_{xy} = \frac{\pi}{2} - \theta'_{xy} \\ \gamma_{yz} = \frac{\pi}{2} - \theta'_{yz} \\ \gamma_{zx} = \frac{\pi}{2} - \theta'_{zx} \end{array} \right. \Rightarrow \left\{ \begin{array}{l} \theta'_{xy} = \frac{\pi}{2} - \gamma_{xy} \\ \theta'_{yz} = \frac{\pi}{2} - \gamma_{yz} \\ \theta'_{zx} = \frac{\pi}{2} - \gamma_{zx} \end{array} \right.$$



p.130

3-3 The State of Strain at a Point

- Normal strain associated with a line in direction n

$$\varepsilon_n = \frac{dn' - dn}{dn} = \frac{d\delta_n}{dn} \quad \text{or} \quad dn' = (1 + \varepsilon_n)dn$$

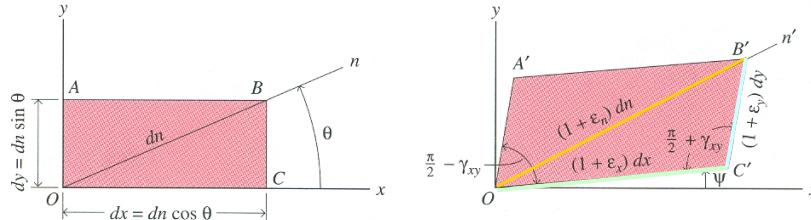
- Shearing strain associated with two orthogonal lines in directions n and t

$$\gamma_{nt} = \frac{\pi}{2} - \theta'_{nt} \quad \text{or} \quad \theta'_{nt} = \frac{\pi}{2} - \gamma_{nt}$$

p.130

3-4 The Strain Transformation Equations for Plane Strain

Plane strain: $\epsilon_z = \gamma_{zx} = \gamma_{zy} = 0$



$$(OB')^2 = (OC')^2 + (C'B')^2 - 2(OC')(C'B')\cos\left(\frac{\pi}{2} + \gamma_{xy}\right)$$

$$[(1 + \epsilon_n)dn]^2 = [(1 + \epsilon_x)dx]^2 + [(1 + \epsilon_y)dy]^2 - 2[(1 + \epsilon_x)dx][(1 + \epsilon_y)dy][-\sin \gamma_{xy}]$$

p.130

Strain Transformation for Normal Strain

$$[(1 + \epsilon_n)dn]^2 = [(1 + \epsilon_x)dx]^2 + [(1 + \epsilon_y)dy]^2 - 2[(1 + \epsilon_x)dx][(1 + \epsilon_y)dy][-\sin \gamma_{xy}]$$

$$dx = dn \cos \theta \quad dy = dn \sin \theta$$

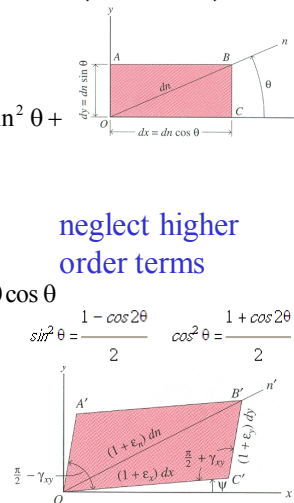
$$(1 + \epsilon_n)^2 (dn)^2 = (1 + \epsilon_x)^2 (dn)^2 \cos^2 \theta + (1 + \epsilon_y)^2 (dn)^2 \sin^2 \theta + 2(dn)^2 \sin \theta \cos \theta (1 + \epsilon_x)(1 + \epsilon_y) \sin \gamma_{xy}$$

For small strains: $\epsilon^2 \ll \epsilon$ $\sin \gamma \cong \gamma$ neglect higher order terms

$$1 + 2\epsilon_n = (1 + 2\epsilon_x)\cos^2 \theta + (1 + 2\epsilon_y)\sin^2 \theta + 2\gamma_{xy} \sin \theta \cos \theta$$

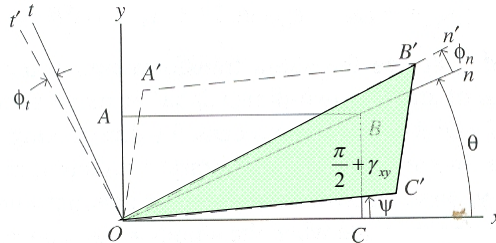
$$\epsilon_n = \epsilon_x \cos^2 \theta + \epsilon_y \sin^2 \theta + \gamma_{xy} \sin \theta \cos \theta$$

$$= \frac{\epsilon_x + \epsilon_y}{2} + \frac{\epsilon_x - \epsilon_y}{2} \cos 2\theta + \frac{\gamma_{xy}}{2} \sin 2\theta$$



p.131

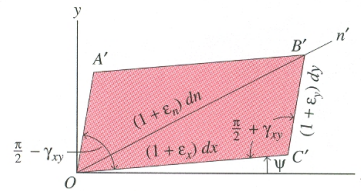
Strain Transformation for **Shearing Strain**



$$\frac{OB'}{\sin \angle OC' B'} = \frac{B' C'}{\sin \angle B' OC'}$$

$$B' C' \sin \angle OC' B' = OB' \sin \angle B' OC'$$

$$(1 + \varepsilon_y) dy \sin \left(\frac{\pi}{2} + \gamma_{xy} \right) = (1 + \varepsilon_n) dn \sin (\theta + \phi_n - \psi)$$



Greek Symbols

p.131

Strain Transformation for **Shearing Strain**

$$(1 + \varepsilon_y) dy \sin \left(\frac{\pi}{2} + \gamma_{xy} \right) = (1 + \varepsilon_n) dn \sin (\theta + (\phi_n - \psi))$$

$$= \cos \gamma_{xy} \cong 1$$

$$\sin (\theta + (\phi_n - \psi)) = \sin \theta \cos (\phi_n - \psi) + \cos \theta \sin (\phi_n - \psi) \cong \sin \theta + (\phi_n - \psi) \cos \theta$$

$$\cong 1 \quad \cong \phi_n - \psi$$

$$(1 + \varepsilon_y) dy \cong (1 + \varepsilon_n) dx [\sin \theta + (\phi_n - \psi) \cos \theta]$$

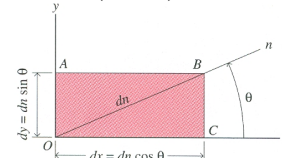
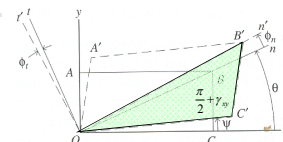
$$= dx \sin \theta$$

$$(\varepsilon_y - \varepsilon_n) \sin \theta \cong (\phi_n - \psi) \cos \theta + \varepsilon_n (\phi_n - \psi) \cos \theta \cong (\phi_n - \psi) \cos \theta$$

$$= (\varepsilon_y - \varepsilon_x \cos^2 \theta - \varepsilon_y \sin^2 \theta - \gamma_{xy} \sin \theta \cos \theta) \sin \theta \quad \sin^2 \theta + \cos^2 \theta = 1$$

$$= -\varepsilon_x \sin \theta \cos^2 \theta + \varepsilon_y \sin \theta \cos^2 \theta - \gamma_{xy} \sin^2 \theta \cos \theta$$

$$\phi_n = -(\varepsilon_x - \varepsilon_y) \sin \theta \cos \theta - \gamma_{xy} \sin^2 \theta + \psi$$



p.132

Strain Transformation for **Shearing Strain**

$$\phi_n = -(\epsilon_x - \epsilon_y) \sin \theta \cos \theta - \gamma_{xy} \sin^2 \theta + \psi$$

$$\phi_n = \phi(\theta)$$

$$\phi_t = \phi\left(\theta + \frac{\pi}{2}\right)$$

$$= -(\epsilon_x - \epsilon_y) \sin\left(\theta + \frac{\pi}{2}\right) \cos\left(\theta + \frac{\pi}{2}\right) - \gamma_{xy} \sin^2\left(\theta + \frac{\pi}{2}\right) + \psi$$

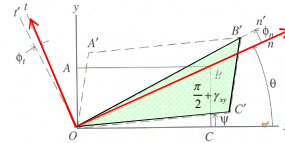
$$= (\epsilon_x - \epsilon_y) \cos \theta \sin \theta - \gamma_{xy} \cos^2 \theta + \psi$$

$$\gamma_{nt} = \phi_n - \phi_t$$

$$= -2(\epsilon_x - \epsilon_y) \cos \theta \sin \theta + \gamma_{xy} (\cos^2 \theta - \sin^2 \theta)$$

$$= -(\epsilon_x - \epsilon_y) \sin 2\theta + \gamma_{xy} \cos 2\theta$$

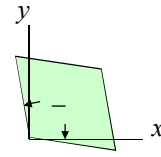
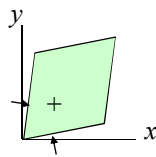
$$\gamma_{nt} = -(\epsilon_x - \epsilon_y) \sin 2\theta + \gamma_{xy} \cos 2\theta$$



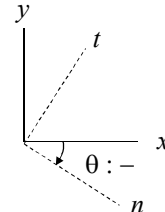
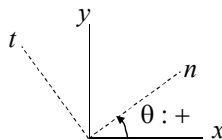
p.133

Sign Conventions

- Tensile strains : +, compressive strains : -
- Shearing strains :



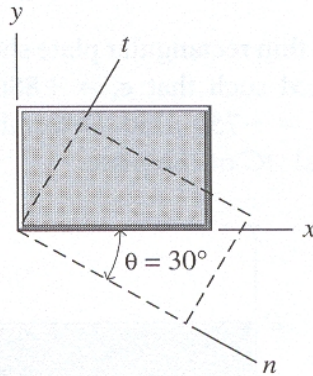
- Angle θ



- (n, t, z) axes have the same order as the (x, y, z) axes.
Both sets right-hand coordinate system.

p.133

Example Problem 3-4



$$\varepsilon_x = +800\mu$$

$$\varepsilon_y = -1000\mu$$

$$\gamma_{xy} = -600\mu$$

$$\theta_n = -30^\circ$$

$$\theta_t = 60^\circ$$

$$\varepsilon_n = ?$$

$$\varepsilon_t = ?$$

$$\gamma_{nt} = ?$$

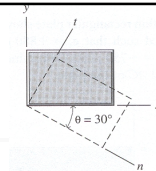
p.133

Example Problem 3-4

$$\varepsilon_x = +800\mu$$

$$\varepsilon_y = -1000\mu$$

$$\gamma_{xy} = -600\mu$$



$$\begin{aligned} \varepsilon_n &= \varepsilon_x \cos^2 \theta + \varepsilon_y \sin^2 \theta + \gamma_{xy} \sin \theta \cos \theta \\ &= 800 \cos^2(-30^\circ) + (-1000) \sin^2(-30^\circ) + (-600) \sin(-30^\circ) \cos(-30^\circ) \\ &= 609.8\mu \cong 610\mu \end{aligned}$$

$$\begin{aligned} \varepsilon_t &= \varepsilon_x \cos^2 \theta + \varepsilon_y \sin^2 \theta + \gamma_{xy} \sin \theta \cos \theta \\ &= 800 \cos^2(+60^\circ) + (-1000) \sin^2(+60^\circ) + (-600) \sin(+60^\circ) \cos(+60^\circ) \\ &= -809.8\mu \cong -810\mu \end{aligned}$$

$$\begin{aligned} \gamma_{nt} &= -2(\varepsilon_x - \varepsilon_y) \cos \theta \sin \theta + \gamma_{xy} (\cos^2 \theta - \sin^2 \theta) \\ &= -2(800 - (-1000)) \cos(-30^\circ) \sin(-30^\circ) + (-600) [\cos^2(-30^\circ) - \sin^2(-30^\circ)] \\ &= 1258.8\mu\text{rad} \cong 1259\mu\text{rad} \end{aligned}$$

Check:

$$\begin{aligned} \gamma_{nt} &= 0.0012588 \\ \sin \gamma_{nt} &= 0.00125900 \cong \gamma_{nt} \\ \cos \gamma_{nt} &= 0.9999992 \cong 1 \end{aligned}$$

Stress vs. Strain Transformation

$$\begin{cases} \sigma_n = \sigma_x \cos^2 \theta + \sigma_y \sin^2 \theta + 2\tau_{xy} \sin \theta \cos \theta \\ \varepsilon_n = \varepsilon_x \cos^2 \theta + \varepsilon_y \sin^2 \theta + \gamma_{xy} \sin \theta \cos \theta \end{cases}$$

$$\begin{cases} \tau_{nt} = -(\sigma_x - \sigma_y) \sin \theta \cos \theta + \tau_{xy} (\cos^2 \theta - \sin^2 \theta) \\ \gamma_{nt} = -2(\varepsilon_x - \varepsilon_y) \cos \theta \sin \theta + \gamma_{xy} (\cos^2 \theta - \sin^2 \theta) \\ \frac{\gamma_{nt}}{2} = -(\varepsilon_x - \varepsilon_y) \cos \theta \sin \theta + \frac{\gamma_{xy}}{2} (\cos^2 \theta - \sin^2 \theta) \end{cases} \quad \Rightarrow$$

$$\begin{aligned} \sigma_n &\leftrightarrow \varepsilon_n \\ \sigma_x &\leftrightarrow \varepsilon_x \\ \sigma_y &\leftrightarrow \varepsilon_y \\ \tau_{xy} &\leftrightarrow \frac{\gamma_{xy}}{2} \\ \tau_{nt} &\leftrightarrow \frac{\gamma_{nt}}{2} \end{aligned}$$

p.135

3-5 Principal Strains and Maximum Shearing Strains

- Similar to the case of plane stress

$$\tan 2\theta_p = \frac{2\tau_{xy}}{\sigma_x - \sigma_y}$$

$$\tan 2\theta_p = \frac{\gamma_{xy}}{\varepsilon_x - \varepsilon_y}$$

$$\sigma_p = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\varepsilon_p = \frac{\varepsilon_x + \varepsilon_y}{2} \pm \sqrt{\left(\frac{\varepsilon_x - \varepsilon_y}{2}\right)^2 + \left(\frac{\gamma_{xy}}{2}\right)^2}$$

$$\tau_p = \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\frac{\gamma_p}{2} = \pm \sqrt{\left(\frac{\varepsilon_x - \varepsilon_y}{2}\right)^2 + \left(\frac{\gamma_{xy}}{2}\right)^2}$$

p.136

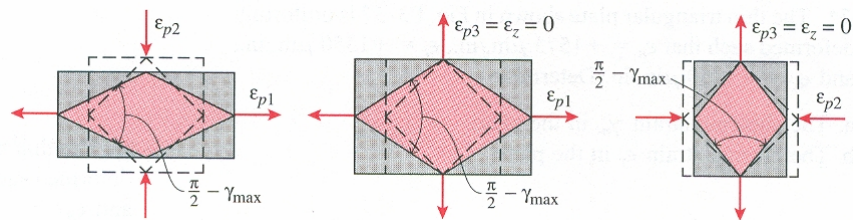
Remarks

- Plane strain:

$$\text{max. } \varepsilon_n : \varepsilon_{p1}, \varepsilon_{p2}, 0$$

$$\text{max. } \gamma_{nt} : \varepsilon_{p1} - \varepsilon_{p2}, \varepsilon_{p1} - 0, \varepsilon_{p2} - 0$$

- The lines associated with the maximum γ_{nt} bisect the angles between the lines of ε_{max} and ε_{min}



p.136

Example Problem 3-5(I)

Given $\varepsilon_x = +1200\mu$ $\varepsilon_y = -600\mu$ $\gamma_{xy} = +900\mu$

:

Find: ■ $\varepsilon_{p1}, \varepsilon_{p2}, \gamma_{\text{max}} = ?$

$$\begin{aligned} \text{Sol: } \varepsilon_{p1}, \varepsilon_{p2} &= \frac{\varepsilon_x + \varepsilon_y}{2} \pm \sqrt{\left(\frac{\varepsilon_x - \varepsilon_y}{2}\right)^2 + \left(\frac{\gamma_{xy}}{2}\right)^2} \\ &= \left[\frac{1200 + (-600)}{2} \pm \sqrt{\left(\frac{1200 - (-600)}{2}\right)^2 + \left(\frac{900}{2}\right)^2} \right] \times 10^{-6} \\ &= [300 \pm 1006.2] \times 10^{-6} \end{aligned}$$

$$\therefore \varepsilon_{p1} \cong 1306\mu \quad \varepsilon_{p2} \cong -706\mu \quad \varepsilon_{p3} = \varepsilon_z = 0$$

p.137

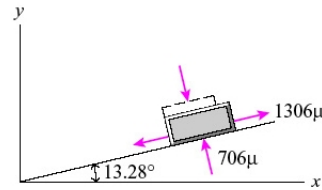
Example Problem 3-5(II)

$$\tan 2\theta_p = \frac{\gamma_{xy}}{\epsilon_x - \epsilon_y} = \frac{900}{1200 - (-600)} = 0.500$$

$$\theta_{p1} = 13.285^\circ \cong 13.28^\circ \rightarrow \epsilon_{p1} = 1306 \mu$$

$$\theta_{p2} = \theta_{p1} + 90^\circ \cong 103.28^\circ \rightarrow \epsilon_{p2} = -706 \mu$$

$$\begin{aligned} \epsilon_n &= \epsilon_x \cos^2 \theta + \epsilon_y \sin^2 \theta + \gamma_{xy} \sin \theta \cos \theta \\ &= 1200 \cdot \cos^2(13.285^\circ) - 600 \cdot \sin^2(13.285^\circ) \\ &\quad + 900 \cdot \sin(13.285^\circ) \cdot \cos(13.285^\circ) \\ &= 1306.2 \mu \end{aligned}$$



p.137

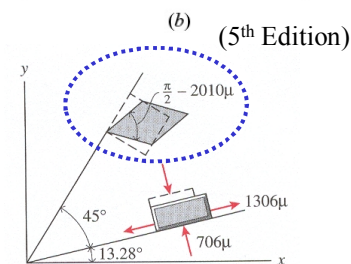
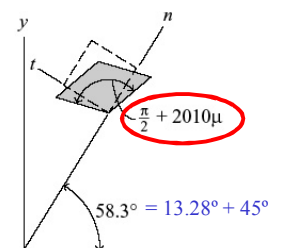
Example Problem 3-5(III)

$$\begin{aligned} \gamma_p &= \pm 2 \sqrt{\left(\frac{\epsilon_x - \epsilon_y}{2}\right)^2 + \left(\frac{\tau_{xy}}{2}\right)^2} \\ &= \pm 2 \sqrt{\left(\frac{1200 - (-600)}{2}\right)^2 + \left(\frac{900}{2}\right)^2} \times 10^{-6} = \pm 2012 \mu \end{aligned}$$

$$\because \epsilon_{p2} < 0 < \epsilon_{p1} \quad \gamma_{\max} = \gamma_p = \pm 2012 \mu \cong \pm 2010 \mu$$

$$\theta_\gamma = 13.285^\circ \pm 45^\circ \cong 58.285^\circ \text{ and } -31.715^\circ$$

$$\begin{aligned} \gamma_{nt} &= -2(\epsilon_x - \epsilon_y) \sin \theta \cos \theta + \gamma_{xy} (\cos^2 \theta - \sin^2 \theta) \\ &= -2(1200 + 600) \sin(58.285^\circ) \cos(58.285^\circ) \\ &\quad + 900 [\cos^2(58.285^\circ) - \sin^2(58.285^\circ)] \\ &= -2012.5 \mu \end{aligned}$$



p.138

Example Problem 3-6 (I)

$$\varepsilon_x = +720\mu \quad \varepsilon_y = +520\mu \quad \gamma_{xy} = +480\mu$$

$$\blacksquare \varepsilon_{p1}, \varepsilon_{p2}, \gamma_{\max} = ?$$

$$\begin{aligned} \varepsilon_{p1}, \varepsilon_{p2} &= \frac{\varepsilon_x + \varepsilon_y}{2} \pm \sqrt{\left(\frac{\varepsilon_x - \varepsilon_y}{2}\right)^2 + \left(\frac{\gamma_{xy}}{2}\right)^2} \\ &= \left[\frac{720 + 520}{2} \pm \sqrt{\left(\frac{720 - 520}{2}\right)^2 + \left(\frac{480}{2}\right)^2} \right] \times 10^{-6} \\ &= [620 \pm 260] \times 10^{-6} \end{aligned}$$

$$\varepsilon_{p1} \cong 880\mu \quad \varepsilon_{p2} \cong 360\mu \quad \varepsilon_{p3} = \varepsilon_z = 0$$

p.138

Example Problem 3-6 (II)

$$\begin{aligned} \gamma_p &= 2\sqrt{\left(\frac{\varepsilon_x - \varepsilon_y}{2}\right)^2 + \left(\frac{\tau_{xy}}{2}\right)^2} \\ &= 2\sqrt{\left(\frac{720 - 520}{2}\right)^2 + \left(\frac{480}{2}\right)^2} \times 10^{-6} = 520\mu \end{aligned}$$

$$\because 0 < \varepsilon_{p2} < \varepsilon_{p1} \quad \varepsilon_{p1} \cong 880\mu \quad \varepsilon_{p2} \cong 360\mu \quad \varepsilon_{p3} = \varepsilon_z = 0$$

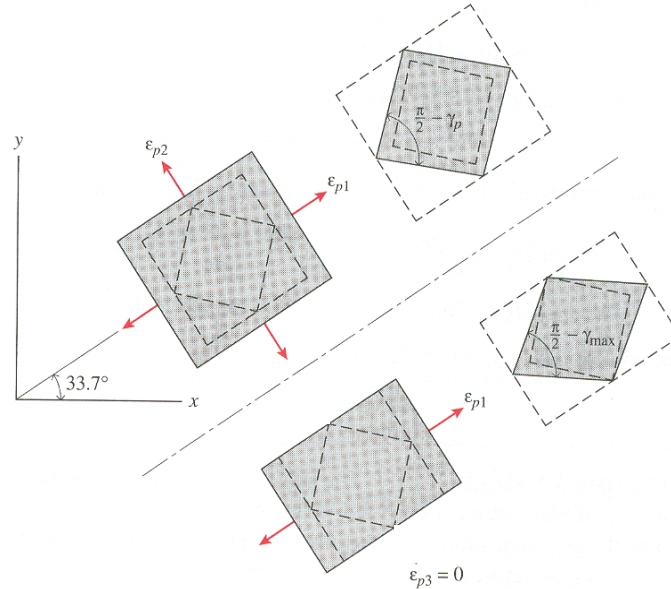
$$\gamma_{\max} = \varepsilon_{p1} - 0 = 880\mu$$

$$\tan 2\theta_p = \frac{\gamma_{xy}}{\varepsilon_x - \varepsilon_y} = \frac{480}{720 - 520} = 2.400$$

$$\theta_p = 33.69^\circ \cong 33.7^\circ$$

p.138

Example Problem 3-6 (III)



p.140

3-6 Mohr's Circle for Plane Strain

$$\left(\sigma_n - \frac{\sigma_x + \sigma_y}{2} \right)^2 + \tau_{nt}^2 = \left(\frac{\sigma_x - \sigma_y}{2} \right)^2 + \tau_{xy}^2$$

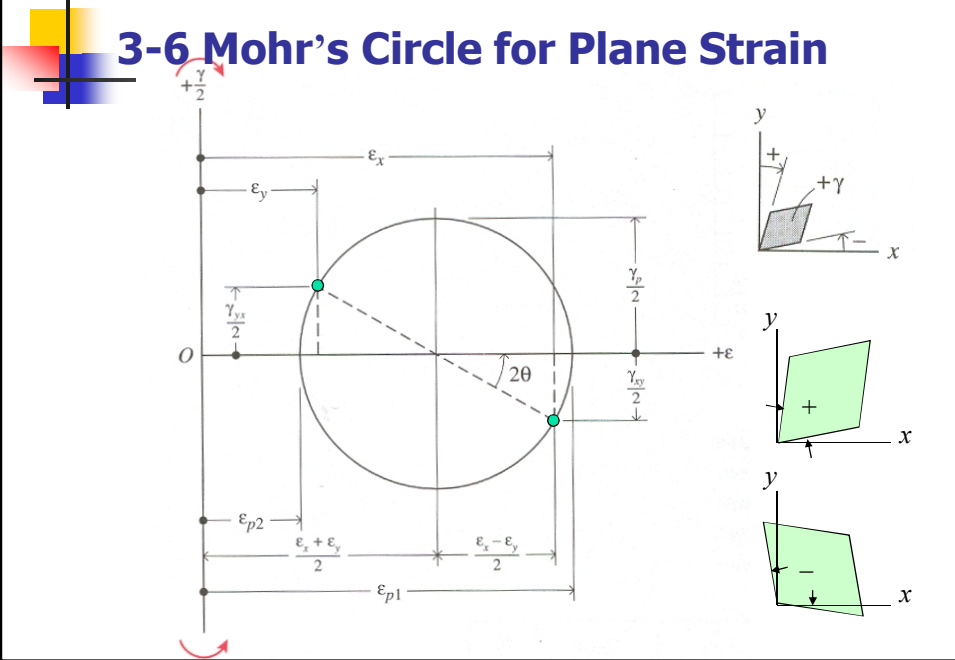
$$\Rightarrow \left(\epsilon_n - \frac{\epsilon_x + \epsilon_y}{2} \right)^2 + \left(\frac{\gamma_{nt}}{2} \right)^2 = \left(\frac{\epsilon_x - \epsilon_y}{2} \right)^2 + \left(\frac{\gamma_{xy}}{2} \right)^2$$

Circle : center $\left(\epsilon_n, \frac{\gamma_{nt}}{2} \right) = \left(\frac{\epsilon_x + \epsilon_y}{2}, 0 \right)$

radius $R = \sqrt{\left(\frac{\epsilon_x - \epsilon_y}{2} \right)^2 + \left(\frac{\gamma_{xy}}{2} \right)^2}$

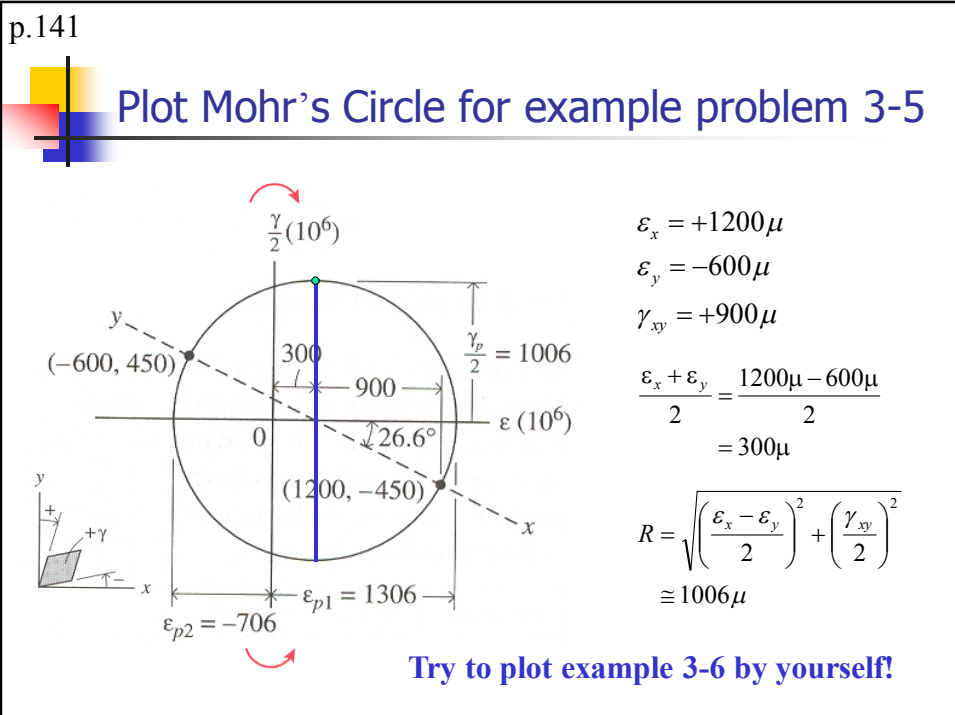
p.141

3-6 Mohr's Circle for Plane Strain



p.141

Plot Mohr's Circle for example problem 3-5



p.142

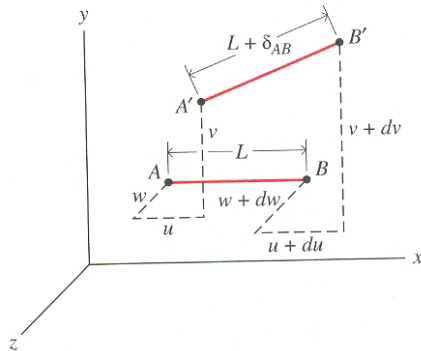
3-7 Strain Measurement and Rosette

- Strain Measurement:

Strain gages **mounted on a free surface**

$$\sigma_z, \tau_{xz}, \tau_{yz} = 0$$

$$\epsilon_x, \epsilon_y, \epsilon_z, \gamma_{xy} \neq 0, \gamma_{zx}, \gamma_{zy} = 0$$



$$(L + \delta_{AB})^2 = (L + du)^2 + (dv)^2 + (dw)^2$$

$$\begin{aligned} & \cancel{L^2} + 2L\delta_{AB} + \delta_{AB}^2 \\ & \approx \cancel{L^2} + 2L(du) + (du)^2 + (dv)^2 + (dw)^2 \end{aligned}$$

neglect 2nd order terms

$$\delta_{AB} = du$$

p.142

3-7 Strain Measurement and Rosette

- Normal strain along $AB = \delta_{AB}/L$ is **not** affected by out-of-plane displacement (dv).

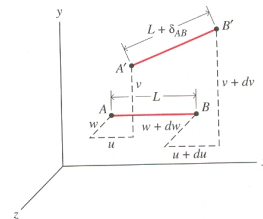
- None** of in-plane strain are affected (dw).

$$\epsilon_n = \frac{\epsilon_x + \epsilon_y}{2} + \frac{\epsilon_x - \epsilon_y}{2} \cos 2\theta + \frac{\gamma_{xy}}{2} \sin 2\theta$$

$$\gamma_{nt} = -(\epsilon_x - \epsilon_y) \sin 2\theta + \gamma_{xy} \cos 2\theta$$

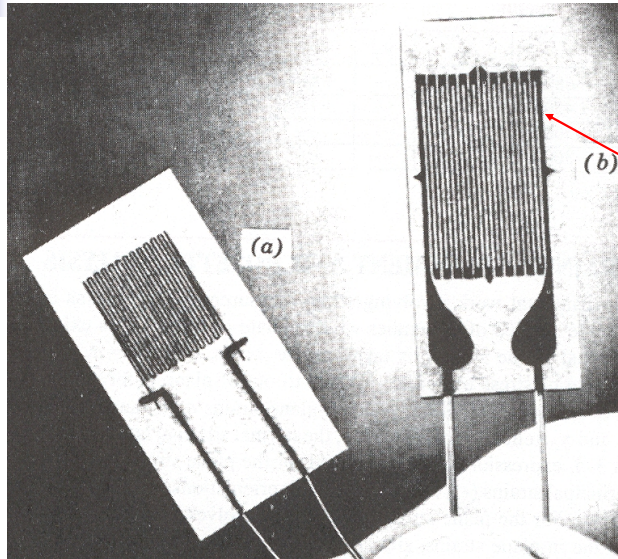
are valid for both plane strain and plane stress states.

- Plane stress: $\epsilon_z = \epsilon_{p3} = -\frac{\nu}{1-\nu}(\epsilon_x + \epsilon_y)$ (section 4-3)



p.143

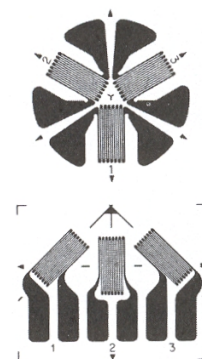
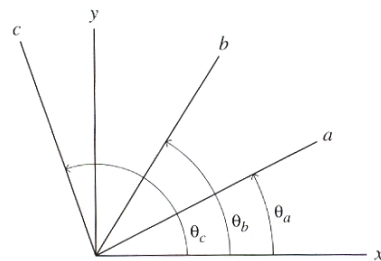
Strain Gage – Measure Normal Strain



0.001 in ϕ
electrical
resistance
changes

p.144

Strain Rosettes (I)



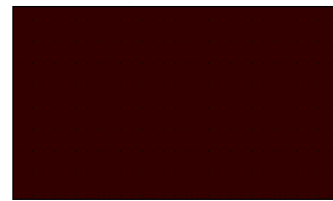
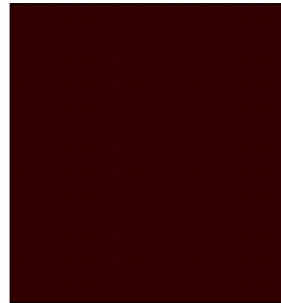
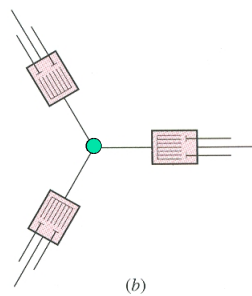
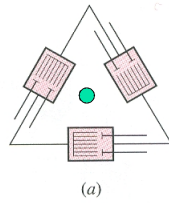
$$\epsilon_a = \epsilon_x \cos^2 \theta_a + \epsilon_y \sin^2 \theta_a + \gamma_{xy} \sin \theta_a \cos \theta_a$$

$$\epsilon_b = \epsilon_x \cos^2 \theta_b + \epsilon_y \sin^2 \theta_b + \gamma_{xy} \sin \theta_b \cos \theta_b \quad \Rightarrow \quad \epsilon_x, \epsilon_y, \gamma_{xy}$$

$$\epsilon_c = \epsilon_x \cos^2 \theta_c + \epsilon_y \sin^2 \theta_c + \gamma_{xy} \sin \theta_c \cos \theta_c$$

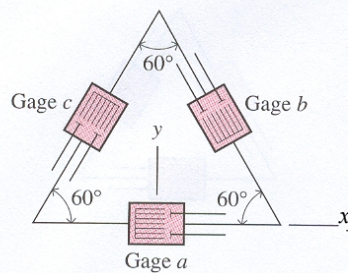
p.144

Strain Rosettes (II)



p.145

Example Problem 3-7 (I)



$$\epsilon_a = +1000\mu$$

$$\epsilon_b = +750\mu$$

$$\epsilon_c = -650\mu$$

$$\epsilon_{p1}, \epsilon_{p2}, \gamma_{\max} = ?$$

Figure 3-19

$$\epsilon_a = \epsilon_0 = +1000\mu \text{ (measured)} = \epsilon_x$$

$$\epsilon_c = \epsilon_{60} = -650\mu = 1000\mu \cos^2 60^\circ + \epsilon_y \sin^2 60^\circ + \gamma_{xy} \sin 60^\circ \cos 60^\circ$$

$$\epsilon_b = \epsilon_{120} = +750\mu = 1000\mu \cos^2 120^\circ + \epsilon_y \sin^2 120^\circ + \gamma_{xy} \sin 120^\circ \cos 120^\circ$$

p.145

Example Problem 3-7 (II)

$$\varepsilon_y = -266.7\mu \quad \gamma_{xy} = -1616.6\mu$$

$$\begin{aligned}\varepsilon_{p1}, \varepsilon_{p2} &= \frac{\varepsilon_x + \varepsilon_y}{2} \pm \sqrt{\left(\frac{\varepsilon_x - \varepsilon_y}{2}\right)^2 + \left(\frac{\gamma_{xy}}{2}\right)^2} \\ &= \left[\frac{1000 - 266.7}{2} \pm \sqrt{\left(\frac{1000 + 266.7}{2}\right)^2 + \left(\frac{-1616.6}{2}\right)^2} \right] \times 10^{-6} \\ &= [366.7 \pm 1026.9] \times 10^{-6}\end{aligned}$$

$$\varepsilon_{p1} \cong 1394\mu \quad \varepsilon_{p2} \cong -660\mu$$

$$\varepsilon_z = \varepsilon_{p3} = -\frac{\nu}{1-\nu}(\varepsilon_x + \varepsilon_y) = -\frac{1/3}{1-1/3}(1000\mu - 266.7\mu) \cong -367\mu$$

p.146

Example Problem 3-7 (III)

$$\therefore \varepsilon_{p2} = -660\mu < \varepsilon_{p3} = -367\mu < \varepsilon_{p1} = 1394\mu$$

$$\gamma_{max} = \gamma_p = \varepsilon_{p1} - \varepsilon_{p2} = 1393.6\mu + 660.2\mu = 2053.8\mu \cong 2050\mu$$

$$\tan 2\theta_p = \frac{\gamma_{xy}}{\varepsilon_x - \varepsilon_y} = \frac{-1616.6}{1000 + 266.7} = -1.2763$$

$$\theta_p = -25.96^\circ \cong -26.0^\circ$$

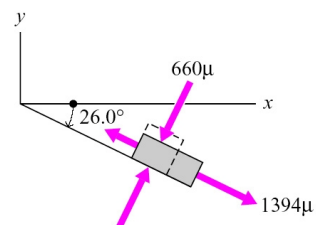


Figure 3-21



8 Exercises

3-10, 3-26, 3-36, 3-45,
3-53, 3-57, 3-66, 3-75